

**N 93 - 22118**

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## ENTRY SYSTEMS BACKGROUND

- **HYPersonic VEHICLES STUDIES**

- Aerothermal / Structural Concepts AFWAL 1985-1987

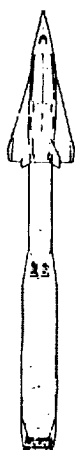
- **AEROBRAKING SPACE TRANSFER VEHICLES (ASTV) STUDIES**

- Concepts Definition studies/ Turnaround Operations/ Space Navigation and Aerobraking/ Centaur- derived Lunar Transfer Vehicles NASA centers 1979-1990
  - ASTV-related IR&D Studies involving wind- tunnel testing, aerothermodynamics, GN&C and STV design studies 1983-1991

## AEROTHERMAL / STRUCTURAL CONCEPTS STUDY

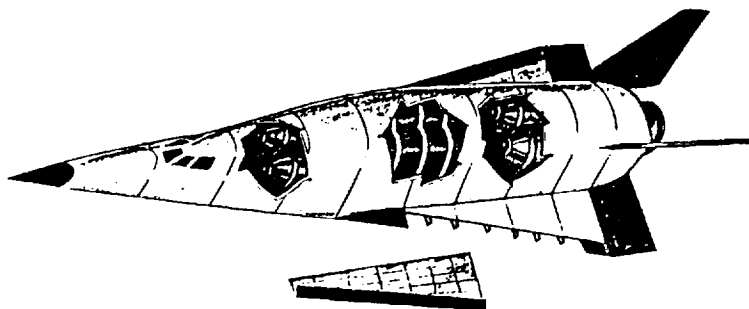
### OBJECTIVES

- Establish aerothermal environments for hypersonic aerospace vehicles.
- Develop thermostructural design concepts.
- Obtain optimum Thermostructural designs by performing trade studies
- Identify areas for further development



Length 95 ft  
Height 26 ft 8 in.  
Wing span 37 ft 6 in.  
Takeoff weight 98,000 lb  
Payload 5,000 lb  
Empty 43,000 lb  
Propellants 48,400 lb  
LO<sub>2</sub> 41,500 lb  
LH<sub>2</sub> 6,900 lb

Suborbital vehicle  
and booster



## TPS TECHNOLOGY REQUIREMENTS

- ADVANCED RADIATORS, INSULATORS AND ABLATORS
  - COATED REFRACTORY METALS
  - RIGID CERAMICS
  - FLEXIBLE CERAMICS
  - ADVANCED CARBON CARBON
- ACTIVE COOLING DEVICES FOR HOT STRUCTURES

## PROGRAM ENABLING TECHNOLOGY ASSESSMENT

Program Area: Hypersonics

Technology Area: Aerothermodynamics

Priority Requirement (Source)	Government Technology Development	Industry Technology Development
<b>Enabling Technology</b>		
Aerodynamic Heating	<u>Current</u> <ul style="list-style-type: none"> <li>• SEI Studies</li> <li>• NASP related studies</li> <li>• HYFLEX</li> </ul>	<u>Current</u>
<b>Enabling Technology</b> Real gas effects Boundary layer transition Turbulence modeling Shock boundary layer interaction Shock impingement Rarefied flows Chemical non-equilibrium Thermal non-equilibrium Surface catalysis/surface reflectance	<u>Needed</u> <ul style="list-style-type: none"> <li>• Validated CFD methods</li> <li>• Ground test (materials) data</li> <li>• Flight test data               <ul style="list-style-type: none"> <li>- HGV flight test</li> <li>- AFE (14' brake)</li> <li>- Deployable AFE (45' brake)</li> </ul> </li> </ul>	<u>Needed</u>
Thermal Control	<u>Current</u>	<u>Current</u>
<b>Enabling Technology</b> High temperature heat pipes Nose-tip and Leading edge cooling/ temperature control Active cooling Antenna cooling Electronics cooling Insulation Ablation	<u>Needed</u>	<u>Needed</u>

# PROGRAM ENABLING TECHNOLOGY ASSESSMENT

Program Area: Hypersonics

Technology Area: High Temperature Structures and TPS

Priority Requirement (Source) Enabling Technology	Government Technology Development	Industry Technology Development
Affordable, Reliable Hot Structures	<u>Current</u>	<u>Current</u>
<b>Enabling Technology</b> High temperature materials Hybrid design Joints, seals and adhesives Nose and leading edge Fasteners	<u>Needed</u>	<u>Needed</u>
High Temperature TPS	<u>Current</u>	<u>Current</u>
<b>Enabling Technology</b> Carbon/carbon insulation High temperature flexible TPS High temperature rigid TPS Active cooling Ablators	<u>Needed</u>	<u>Needed</u>